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HIGHLIGHTS

- For the first time, Smoothed-Particle Hydrodynamics (SPH) has been implemented to predict and assess mixing behaviors in oscillatory baffled reactors.
- Velocity fields and flow patterns modelled by SPH match closely to those by its Eulerian counterpart, i.e. Finite Volume (FV) method.
- Historical information of individually tracked fluid particles, provided by SPH, is directly applied to quantify mixing performance using the existing velocity ratio and stretch rate indexes.
- By making full use of SPH's capabilities, two new indexes are proposed for the quantification of mixing and plug flow efficiency of the device.

Smoothed Particle Hydrodynamics - A New Approach for Modeling Flow in Oscillatory Baffled Reactors

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ABSTRACT

Existing numerical models of mixing processes in oscillatory baffled reactors (OBR) are mainly Eulerian-based. An alternative Lagrangian based methodology, Smoothed-Particle Hydrodynamics (SPH), for predicting flow patterns and assessing mixing performance is presented in this paper. A bespoke SPH solver is developed for single phase modelling, as it is, for the first time, applied to OBR, and the results are compared with those from Eulerian modelling, i.e. Finite Volume (FV) method. SPH has successfully captured the expected flow characteristics in OBR as clearly and equally as its Eulerian counterpart, validating the SPH method. Since SPH provides historical information of individually tracked fluid packets/particles in the domain of interest, it allows for readily quantitative assessments of mixing without additional models. Two new indexes to assess mixing and plug flow efficiency have been proposed by making full use of SPH's capabilities.

Key words: Smoothed Particle Hydrodynamics (SPH); Computational Fluid Dynamics (CFD); Lagrangian based modeling; oscillatory baffled reactor; flow patterns; mixing efficiency

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1. Introduction

Uniform mixing and linear scale up offered by Oscillatory Baffled Reactors (OBR) are the key drivers for research and industrial applications in reactions (Eze et al., 2017; Navarro Fuentes et al., 2018; Ni et al., 1999; Phan et al., 2012; Wilson et al., 2001) and crystallization (Agnew et al., 2017; Brown et al., 2015; Brown et al., 2018; Brown & Ni, 2012; Brown & Ni, 2011; Callahan & Ni, 2012; Feilden, 2017; Lawton et al., 2009; Ni et al., 2004; Su et al., 2015). From a modelling viewpoint, Computational Fluid Dynamic (CFD) solvers have been employed to simulate hydrodynamic flow profiles in OBR from initially 2-dimensional in the 1990s (Howes et al., 1991; Howes & Shardlow, 1997; Mackley & Neves Saraiva, 1999; Mackley & Roberts, 1991; Roberts, 1994; Roberts & Mackley, 1995, 1996) to 3-D in the following decades (Chew et al., 2004; Deng et al., 2010; Fitch et al., 2005; Hamzah et al., 2012; Jian & Ni, 2003; Jian & Ni, 2005; Mazubert et al., 2016a; Ni et al., 2003; Ni et al., 2002; Nogueira et al., 2013; Reis et al., 2005; Reis et al., 2010; Zheng et al., 2007), however the majority of the numerical modelling work was Eulerian based (González-Juárez et al., 2017; Manninen et al., 2013) mostly for single phase. The validation of these numerical models with experimental data has mostly been limited to qualitative comparison of eddy formation patterns. There were few modelling work involving two phases, e.g. solids and liquid (Mazubert et al., 2016b), which is largely limited by the expensive computational costs of coupling continuous Eulerian-Lagrangian phases for dynamic fluid-solids modelling and their complex interactions. Coupling CFD with the Population Balance Equation (PBE) to model crystallization processes in stirred tank crystallizers is a well-documented example (Yang & Mao, 2014). The question remains if the complexity of these coupled methodologies can be avoided and if all phases can be modeled using a Lagrangian scheme; the Smoothed-Particle Hydrodynamics (SPH) is thus the one considered in the current study. The SPH methodology provides historical information of individual fluid packets or particles in the

domain of interest by tracking these particles; this implicit SPH capability opens up opportunities for better understanding of flow rheology behavior. This is of special interest in complex processes where L-L and S-L interactions play a key role, such as the chemical processes commonly undertaken in OBRs, e.g. crystallization. This can greatly improve the accuracy in the prediction of mixing, since historical information of particles allows for alternative ways to quantify mixing, as opposed to residence time distributions and axial dispersion assessments traditionally used in OBRs. The quantification of mixing efficiency in OBRs is of great interest and has been a hot topic of research since the 1990s in both experimental (Mackley & Ni, 1993; Ni, 1994; Ni, 1995; Ni & Stevenson, 1999; Palma & Giudici, 2003; Phan & Harvey, 2010; Phan & Harvey, 2011; Reis et al., 2004; Zheng & Mackley, 2008) and numerical studies (González-Juárez et al., 2018; González-Juárez et al., 2017; Howes et al., 1991; Jian & Ni, 2003; Kimuli et al., 2017; Manninen et al., 2013; Reis et al., 2010), since both uniform mixing and heat transfer control in OBRs are the essential elements for achieving consistent product properties in these undertaken processes, e.g. narrow crystal size distribution, constant polymorph. In addition, SPH allows for the implementation of new physics based on particle-particle interaction, which has great potential for modelling solids formation and growth in crystallization processes.

SPH was first developed in 1977 for astrophysical applications by Gingold and Monaghan (Gingold & Monaghan, 1977) and its application into fluid flow problems was proposed by Monaghan in the early 1990s (Monaghan, 1992). This methodology, despite being less popular than traditional standard CFD approaches, is well documented (Violeau, 2012), and has augmented vast research attention in recent years in solving complex fluid flow problems (Adami et al., 2013; Crespo et al., 2007; Dalrymple & Rogers, 2006; Ferrari et al., 2009; Gomez-Gesteira et al., 2010; Zhang et al., 2017) in micro and macro-scales. To the authors'

knowledge there has been no reported work on the application of SPH for modelling fluid flows in tubular baffled reactors. The aims of this work are to explore and investigate the feasibility of SPH in OBR by developing a bespoke solver; to compare flow characteristics of single phase flow predicted by SPH with those obtained with an Eulerian based model, e.g. Finite Volume (FV), via a commercial software package (ANSYS® Fluent 15) and to provide quantitative assessment of mixing efficiency using the proposed methodology.

2. Problem definition

Oscillatory baffled reactors are tubular devices that contain periodically spaced orifice baffles with superimposed fluid oscillation; mixing is generated by the formation and cessation of eddies, providing uniform mixing in each inter-baffled section (Jian et al., 2004; Mackley & Ni, 1991). The sinusoidal oscillatory inlet flow is imposed by a piston at the entrance of the reactor. The displacement of the piston and the oscillatory inlet velocity of the reactor are represented by:

$$x_p(t) = -x_o \cos(\omega t) \quad (1)$$

$$u_{inlet}(t) = \omega x_o \sin(\omega t) \quad (2)$$

where x_p is the piston's position (m), u_{inlet} the inlet mean velocity (m s^{-1}), $\omega = 2\pi f$ the oscillation angular frequency (rad s^{-1}), f the oscillation frequency (Hz), x_o the oscillation center-to-peak amplitude (m) and t the time (s). Figure 1 and Table 1 show the schematic and geometry of the oscillatory baffled reactor system.

Figure 1. Schematic of the baffled cell and inlet velocity profile in OBR; all quotes are in mm

Table 1. Geometry dimensions of the OBR

D (mm)	15
D_b (mm)	4
L_b (mm)	7
L (mm)	24
Baffled cells in reactor (#)	2

where D is the diameter of the tube (m), D_b the diameter of baffle hole (m), L_b the baffle spacing (m) and L the length of the reactor (m). Water is used as the working fluid with oscillatory conditions of $x_o = 5$ mm and $f = 1$ Hz ($Re_o = \omega x_o \rho D / \mu = 471$ and $St = D / (4\pi x_o) = 0.239$); these conditions are chosen to avoid interactions among eddies generated during the forward and backward strokes so that symmetrical flow patterns are expected.

3. Numerical formulation

The simulated system comprises a 2-dimensional domain and it is assumed iso-thermal.

Navier-Stokes equations for mass and momentum conservation are solved according to:

$$\frac{D\rho}{Dt} + \rho \nabla \cdot \vec{u} = 0 \quad (3)$$

$$\frac{D(\rho \vec{u})}{Dt} = \nabla \cdot \underline{\underline{\sigma}} + \rho \vec{g} + \sum \vec{F} \quad (4)$$

where $\frac{D}{Dt}$ is the material derivative, ∇ the gradient vector operator, ρ is the density, $\vec{u}$ the

velocity vector, $\underline{\underline{\sigma}}$ the total stress tensor defined as $\underline{\underline{\sigma}} = -p\underline{\underline{I}} + \underline{\underline{\tau}}$ and $\sum \vec{F}$ the summation of any external forces. Within the stress tensor, $\underline{\underline{I}}$ is the identity matrix, p the pressure,

$\underline{\underline{\tau}} = \left(-\frac{2}{3} \mu \nabla \cdot \vec{u} \right) \underline{\underline{I}} + 2\mu \underline{\underline{E}}$ is the stress tensor with $\underline{\underline{E}}$, the strain rate of tensor, given by

$\underline{\underline{E}} = \frac{1}{2} \left[(\nabla \vec{u}) + (\nabla \vec{u})^T \right]$. Equations (3) and (4) are discretized and solved separated using SPH

and FV in their respective solvers. Note that the governing equations have been solved in a 2-dimensional form, neither for FV nor SPH were these solved axisymmetrically.

3.1. Smoothed-Particle Hydrodynamic

SPH discretizes the fluid domain using a fixed number of infinitesimal particles that are tracked individually. Assuming that the system is fully flooded with a single-phase fluid, no external forces are considered; hence Equations (3) and (4) can be written in their Lagrangian form as:

$$\frac{D\rho}{Dt} = -\rho \nabla \cdot \vec{u} \quad (5)$$

$$\frac{D\vec{u}}{Dt} = -\frac{1}{\rho} \nabla \cdot p \underline{I} + \frac{1}{\rho} \nabla \cdot \underline{\underline{\tau}} \quad (6)$$

and the position of each infinitesimal particle $\vec{r}$ is governed by:

$$\frac{d\vec{r}}{dt} = \vec{u} \quad (7)$$

This system of equations is closed using the relationship between density and pressure. Here, the fluid domain is treated as weakly compressible, for which this relationship is given by the Equation-Of-State (EOS) (Monaghan, 1992). The most common EOS was proposed by Batchelor (Batchelor, 1974), the so-called *Tait's* equation:

$$p = B \left[\left(\frac{\rho}{\rho_0} \right)^\gamma - 1 \right] \quad (8)$$

where γ usually takes a value of 7 for water, ρ_0 is the reference density of the fluid at atmospheric pressure and the constant B is defined as $B = \frac{\rho_0 c_0^2}{\gamma}$; c_0 is the reference speed of sound. The use of *Tait's* equation as EOS highly reduces computational time, as oppose to solving the Poisson's equation for incompressible flows (Lee et al., 2007). The compressibility of the fluid in the SPH's weakly compressible approach is controlled by c_0 , which also determines the size of the time-step based on the Courant–Friedrichs–Lewy condition (Courant et al., 1928), refer to Equation (15). High values of speed of sound result

in both incompressible behavior and very small time-steps. However, Monaghan (Monaghan, 1994) demonstrated that c_0 values of ten to a hundred times the maximum velocity in the system, corresponding to a Mach number of less than 0.1, replicate incompressible flow with density fluctuations within 1%. A value of $c_0 = 10 \text{ m s}^{-1}$ was then chosen for the simulations performed in this study.

3.1.1. Discretization and kernel approximation

SPH treats a continuous medium as a discrete set of particles. At time zero, particles are uniformly distributed and are equidistantly spaced from each other, i.e. a distance Δx in all directions. Hence, all particles have an initial volume of Δx^n where n is the dimensions of the domain. The discretization of Equations (5) and (6) is derived from the Dirac distribution δ . Thus, the value of a function at a certain time, t , and at an arbitrary point or particle, i , can then be re-written as the summation the contributions from its neighboring particles, j :

$$f(\vec{r}_i, t) \approx \sum_{j=1}^N \frac{m_j}{\rho_j} f(\vec{r}_j, t) W(\vec{r}_i - \vec{r}_j, h) \quad (9)$$

where f is a function of the position vectors $\vec{r}_i$ and $\vec{r}_j$ – the former refers to the particle of interest and the latter to each of its neighbouring particles. The contribution of each neighboring particle is weighted by the use of a kernel function, W , based on the distance between particles i and j ; h is the smoothing length of the chosen kernel function defined as $h = \eta \Delta x$; η typically takes a value of 1.2 – 1.3, the latter is chosen for this work. From this point onwards, for simplicity purposes $W(\vec{r}_i - \vec{r}_j, h)$ will be denoted as W_{ij} and $(\vec{r}_i - \vec{r}_j)$ as $\vec{r}_{ij}$; N is the number of neighboring particles within the kernel domain.

The kernel function W_{ij} for smoothing length h depends on the magnitude of the position vector $\vec{r}_{ij}$ as:

$$W_{ij} = \frac{\alpha_{w,n}}{h^n} f(q) \quad (10)$$

where $\alpha_{w,n}$ is the kernel renormalization term that depends on the kernel and dimensionality of the problem, $f(q)$ represents the kernel function that is a positive, symmetric and at least once continuously derivable function with $q = \frac{|\vec{r}_{ij}|}{h}$. In this study, the Wendland kernel is employed, as it provides a high order of interpolation with reasonable computational cost (Gomez-Gesteira et al., 2010); a high order of interpolation is required to capture complex fluid flow phenomena, leading to high degrees of accuracy and stability of the SPH scheme (Morris, 1996). The Wendland kernel (Wendland, 1995) is defined as:

$$f(q) = \begin{cases} \left(1 - \frac{q}{2}\right)^4 (1 + 2q) & 0 \leq q \leq 2 \\ 0 & 2 < q \end{cases} \quad (11)$$

where $\alpha_{w,1} = \frac{3}{4}$, $\alpha_{w,2} = \frac{7}{4\pi}$, $\alpha_{w,3} = \frac{21}{16\pi}$ for 1, 2 and 3 dimensions respectively.

3.1.2. Continuity equation

The continuity equation, in its SPH formulation, given in Equation (5), is then discretized and re-written as (Gingold & Monaghan, 1977):

$$\frac{D\rho_i}{Dt} = \sum_{j=1}^N m_j \vec{u}_{ij} \cdot \nabla_i W_{ij} \quad (12)$$

where i is the interested particle, j any neighboring particle within the kernel domain, i.e. $|\vec{r}_{ij}| < 2h$ for the Wendland kernel, ∇_i the gradient of the kernel function with respect to the position vector $\vec{r}_{ij}$ and $\vec{u}_{ij} = \vec{u}_i - \vec{u}_j$ is the relative velocity between the particles.

3.1.3. Momentum equation

The momentum equation, in its SPH formulation, given in Equation (6), is discretized following the discretized form proposed by Morris (Morris et al., 1997):

$$\frac{D\vec{u}_i}{Dt} = - \sum_{j=1}^N m_j \left(\frac{p_j}{\rho_j^2} + \frac{p_i}{\rho_i^2} \right) \nabla_i W_{ij} + \sum_{j=1}^N m_j \left(\frac{\mu_i + \mu_j}{\rho_i \rho_j} \right) \vec{u}_{ij} \left(\frac{1}{|\vec{r}_{ij}|} \frac{\partial W_{ij}}{\partial |\vec{r}_{ij}|} \right) \quad (13)$$

3.1.4. Density-Smoothing method

Equations (12) and (13) are solved and regarded as the standard weakly compressible SPH method. However, while kinematics in SPH is well understood, the weakly compressible approach can result in pressure fluctuations between particles, translating into numerical noise in the velocity field, which may exacerbate disorder and erratic motions of the fluid particles. One of the most straightforward and computationally least expensive approaches to tackle this issue is to perform a smoothing filter over the density of the particles (Gomez-Gesteira et al., 2010). In the past, this has commonly been done by re-assigning a reference density value to each particle at set time intervals (Belytschko et al., 1998; Colagrossi & Landrini, 2003; Dilts, 1999). A more elegant approach, utilized in this work, is to implement

a simple density-smoothing model, analogous to the α -XSPH model as proposed by Violeau (Violeau, 2012):

$$\hat{\rho}_i = \rho_i - \varepsilon \sum_{j=1}^N m_j \frac{\rho_{ij}}{\bar{\rho}_{ij}} W_{ij} \quad (14)$$

where ε is a dimensionless coefficient, $\rho_{ij} = \rho_i - \rho_j$ and $\bar{\rho}_{ij}$ is the harmonic average. Values of the order of 10^{-2} are often recommended for the constant ε of the density-smoothing function (Violeau, 2012); $\varepsilon = 0.01$ is chosen in the present study. When this density-smoothing method is implemented, Equation (14) is solved immediately after solving Equation (12).

3.1.5. Time integration

An adaptive time-stepping algorithm is used for calculating time-steps Δt . Here, three criteria are used which include the Courant–Friedrichs–Lewy condition (Courant et al., 1928):

$$\Delta t_{CFL} \leq C_{CFL} \frac{h}{c_0} \quad (15)$$

a constrain based on the force per unit mass of each particle (Monaghan, 1992), which is essentially the magnitude of particle acceleration $|\vec{f}_i|$:

$$\Delta t_{force} \leq C_{force} \min_i \left(\sqrt{\frac{h}{|\vec{f}_i|}} \right) \quad (16)$$

and an additional constrain due to viscous diffusion (Morris et al., 1997):

$$\Delta t_{visc} \leq C_{visc} \min_i \left(\frac{h^2}{\nu_i} \right) \quad (17)$$

where the kinematic viscosity $\nu_i = \mu_0 / \rho_i$ and μ_0 is the reference viscosity of the fluid. Values for C_{CFL} , C_{force} and C_{visc} of 0.01, 0.0125 and 0.0125, respectively, are conservatively chosen to ensure stability of the solution. The final time-step was chosen as the minimum of these three conditions:

$$\Delta t \leq \min(\Delta t_{CFL}, \Delta t_{force}, \Delta t_{visc}) \quad (18)$$

Time was integrated explicitly using the second order accurate Verlet algorithm (Verlet, 1967):

$$\rho_{i,n+1} = \rho_{i,n-1} + (\Delta t_{n-1} + \Delta t_n) \left(\frac{D\rho_i}{Dt} \right)_n \quad (19)$$

$$\vec{u}_{i,n+1} = \vec{u}_{i,n-1} + (\Delta t_{n-1} + \Delta t_n) \left(\frac{D\vec{u}_i}{Dt} \right)_n \quad (20)$$

$$\vec{r}_{i,n+1} = \vec{r}_{i,n} + \vec{u}_{i,n} \Delta t_n + \frac{\Delta t_n^2}{2} \left(\frac{D\vec{u}_i}{Dt} \right)_n \quad (21)$$

where the past, current and future temporal steps are represented by $n-1$, n and $n+1$, respectively. Note that the use of Equations (19) and (20) will eventually lead to a decoupled system. In order to prevent the solution at odd and even time-steps from diverging, an Euler upwind integration is performed at every M time-steps; a value of M of 50 is presently used.

3.1.6. Boundary conditions

The walls that define the limits of the fluid domain were simulated using dynamic boundary conditions (Crespo et al., 2007), chosen for its computational simplicity. Particles comprising dynamic boundaries are solved like any other fluid particles following the solution of Equations (8), (12) and (13). However, the particle positions remain fixed and are not updated in subsequent iterations in time. In order to ensure all fluid particles have consistently the same number of neighboring particles at all times, solid wall boundaries were modeled with three consecutive rows of dynamic particles.

In order to model the oscillatory behavior of the fluid, two pistons, one on the left and another one on the right, are defined on both ends of the OBR, as shown in Figure 2, using dynamic particles whose positions and velocities are controlled by Equations (1) and (2).

Figure 2. SPH OBR model for $\Delta x = 0.0005m$

3.1.7. Post-processing monitors

The velocity profiles of the flow in the OBR are evaluated along three different lines as depicted in Figure 3: a vertical line at the middle of the left baffle (Line 1) – the section experiencing the highest velocity; a centered vertical line (Line 2) – the area where the

strongest eddy dissipation occurs; and a center horizontal line (Line 3) – the area with the weakest eddy interaction as aforementioned. The velocity magnitude u_m at each point of a line, defined at equidistance intervals of Δx along the monitor lines, is calculated analogously to Equation (9), given by:

$$u_{m,i} = \sum_{j=1}^N \frac{m_j}{\rho_j} u_{m,i} W_{ij} \quad (22)$$

where i is the interpolated monitor point that utilizes information from surrounding particles within the kernel smoothing length.

Figure 3. Position of lines at which results are extracted and evaluated

3.2. Finite volume method

For the purpose of comparison, the Eulerian based Finite Volume (FV) method is also applied to the same geometry and operational conditions as given in Table 1 and Figure 1. The following Navier Stokes for continuity and momentum conservation equations, given by Equations (3) and (4), are simplified and solved for incompressible flow:

$$\nabla \cdot \vec{u} = 0 \quad (23)$$

$$\rho \left(\frac{\partial \vec{u}}{\partial t} + \vec{u} \cdot \nabla \vec{u} \right) = -\nabla p + \mu \nabla^2 \vec{u} \quad (24)$$

3.2.1. Numerical Setup

All the numerical FV modelling is performed using the ANSYS® Fluent 15.0 CFD package. Simulations are done by the pressure-based segregated solver, using the SIMPLE pressure-velocity coupling algorithm. A *second order upwind* scheme was utilized for the spatial discretization of the momentum equation; a *second order* scheme for the interpolation of pressure at the faces of the grid and a *second order implicit* scheme for time discretization. The time-step employed is 0.001 seconds throughout all simulations and the convergence criteria is set to residuals of 10^{-5} for solving both Equations (23) and (24) to ensure accuracy. The average value of the Courant–Friedrichs–Lewy (CFL) coefficient was kept below 0.6 and the maximum CLF value below 4.1.

To replicate the OBR with the double piston in the SPH approach, flat velocity profiles are defined on both ends of the OBR using a customized User Defined Function (UDF) in ANSYS® Fluent given by Equation (2); walls were model as *no-slip* boundaries. Equations (23) and (24) were solved as laminar, i.e. no turbulence model was implemented. This is in agreement with literature, where laminar solver has been the norm for simulating flows in oscillatory baffled reactors in the past (Fitch et al., 2005; Ni et al., 2002; Reis et al., 2005) and present decade (González-Juárez et al., 2018; González-Juárez et al., 2017; Kimuli et al., 2017; Mazubert et al., 2016a, 2016b) under a FV framework, including flows with Re_o values up to 8043 (Jian & Ni, 2005; Manninen et al., 2013).

3.2.2. Mesh

A mesh sensitivity analysis is performed using a selection of computational nodes per baffled cell. Meshes of the two baffled-cell domain of five different resolutions are analyzed for ten oscillatory cycles. Profiles of velocity magnitude extracted at Lines 1, 2 and 3, as shown in Figure 3, are then cycle-averaged over all ten oscillatory cycles and compared for strokes 1 to 4 (refer to Figure 1) for various mesh densities. The coefficient of determination, R^2 , was selected for comparison between meshes, defined by:

$$R^2 = 1 - \frac{\sum_{b=1}^k (\phi_{a,b} - \phi_{1,b})^2}{\sum_{b=1}^k (\phi_{a,b} - \phi_{1,b})^2} \quad (25)$$

where b and k are, respectively, a single data point and the total number of data points of a profile, a is the index of a certain mesh and ϕ the property under evaluation. A summary of the mesh independency analysis is shown in Table 2, clearly indicating that the resolution of mesh #2 is the desired choice on the balance between accuracy and computation time, which is selected for this work.

Figure 4 illustrates the resolution and distribution of mesh #2, which is generated using ANSYS® ICEM containing only hexahedral elements.

Figure 4. Finite volume chosen mesh (#2): 17k nodes per baffled cell**Table 2.** Mesh sensitivity analysis results (R^2) where results from mesh #1 are used as the based for comparison

Line 1					
Mesh #	# Nodes [*]	Stroke 1	Stroke 2	Stroke 3	Stroke 4
1	38 k	–	–	–	–
2	17 k	0.999	0.997	0.999	0.945
3	10 k	0.999	0.997	0.997	0.896
4	6 k	0.994	0.989	0.999	0.948
5	3 k	0.982	0.952	0.993	0.953
Line 2					
Mesh #	# Nodes [*]	Stroke 1	Stroke 2	Stroke 3	Stroke 4
1	38 k	–	–	–	–
2	17 k	0.988	0.988	0.966	0.972
3	10 k	0.986	0.994	0.948	0.933
4	6 k	0.968	0.925	0.893	0.867
5	3 k	0.971	0.760	0.918	0.621
Line 3					
Mesh #	# Nodes [*]	Stroke 1	Stroke 2	Stroke 3	Stroke 4
1	38 k	–	–	–	–
2	17 k	0.997	0.983	0.994	0.983
3	10 k	0.996	0.989	0.989	0.962
4	6 k	0.984	0.978	0.975	0.965
5	3 k	0.924	0.952	0.911	0.922

^{*}The values of number of nodes are per baffled cell.

4. Results and discussions

4.1. Profile development

Previous CFD work on oscillatory baffled reactors by Jian (Jian, 2002) showed that flow patterns in an OBR become repeatable and achieved a quasi-steady-state after 5-7 oscillation

cycles. Following the same methodology, the volume-weighted averaged strain rate as a function of time is shown in Figure 5, displaying a similar quasi-steady state after cycle 4. In this work, cycle-averaged parameters generated between cycles 6 to 10 are used for comparison.

Figure 5. *Convergence of strain rate with time in Fluent*

4.2. Sensitivity test of SPH particles

Figure 6 shows the cycle-averaged velocity magnitude profiles of Line 3 (see Figure 3), generated by SPH, at strokes 1 and 3 (see Figure 1) for three different resolutions in terms of the initial particle distribution spacing, Δx . Strokes 1 and 3 were chosen because the flow at these instances experiences the maximum local velocities, leading to potential lower rates of convergence. A clear convergence in the cycle-averaged velocities is observed with decreasing Δx , and the results become resolution independent at $\Delta x = 0.00025$ m, which is therefore chosen for the investigation.

Figure 6. SPH cycle-averaged velocity magnitude profiles of Line 3 for different Δx values

4.3. Flow patterns and velocity profiles

4.3.1. Flow patterns

Figure 7 shows the cycle-averaged flow patterns modelled by both the FV and the SPH with an additional density-smoothing function (SPH D-S for short) at strokes 1 to 4 of an oscillation cycle (see Figure 1). Both methods predict similar eddy formation throughout the oscillation, leading to good comparison. The subtle differences between the two models are the relative size and the intensity (by color) of eddies. The intensity of flow restriction in strokes 1 and 3 is higher for SPH D-S than that for FV, resulting in larger recirculating velocities at the top and bottom of the baffled cell, causing the eddy structures generated during strokes 2 and 4 to remain closer to the baffle's walls. On the contrary, results generated by the FV have slightly larger velocities along the center of the baffled domain, enhancing a small displacement of eddies towards the center of the baffled cell. This slight difference in eddy displacement is reflected in the velocity profiles presented in the next subsection and the maximum velocities listed in Table 3. Flow patterns observed during strokes 1 and 2 and eddy structures formed in strokes 2 and 4 are of high resemblance for the two

methodologies nonetheless. Overall, the results show that the SPH D-S is a viable method in modelling flows in OBR.

Figure 7. Cycle-averaged velocity magnitude contours at strokes 1 to 4 for FV and SPH D-S

Table 3. u_{max} for different strokes and methodologies

	Strokes 1 & 3	Strokes 2 & 4
Finite Volume Method	0.1476 m s^{-1}	0.0897 m s^{-1}
SPH with Density-Smoothing model	0.1127 m s^{-1}	0.0531 m s^{-1}

4.3.2. Velocity profiles

The cycle-averaged velocity magnitude profiles at Lines 1, 2 and 3 computed by the FV, the standard SPH and SPH D-S methodologies are shown in Figures 8 – 10, respectively; the

density-smoothing function was added in a separate simulation to assess its effect individually.

Figure 7 neatly displays how two noticeable high velocity regions (red color) occur through the baffle constriction during strokes 1 and 3, which is shown in Figure 8 as a double-peak velocity profile across Line 1. The velocities obtained with the SPH methodology alone do not display the expected double-peak pattern as the FV did in the forward (1) and reverse (3) strokes, but exhibit parabolic-like characteristics which indicate excessive localized density fluctuations due to the weakly compressible limitation across the constriction of the baffle. This effect is then minimized when the density-smoothing (D-S) function is introduced to limit the “noise” in density distribution. In doing so, the double-peak velocity profiles are reproduced during strokes 1 and 3, highly resembling those modelled by the FV method. Figure 7 also provides qualitative evidence of how, during strokes 2 and 4, the changes in direction of the flow lead to local stalling across the baffle constriction, which translates into lower and more uniform velocity components in the baffle constriction area as shown in Figure 8. The results from the SPH simulation echo the above nonetheless; again the results from the SPH D-S improve the accuracy of the velocity field and are similar to those provided by the FV method.

For the given geometry and operating conditions, symmetrical-mirrored flow patterns and similar velocity magnitudes between the peak and trough are expected (see Figure 7). The velocity profiles predicted by SPH D-S (in Figure 8) at strokes 1 and 3 are indeed very similar in both shape and magnitude with a maximum difference of 4.7%, while the results computed by the FV approach at the same strokes have a larger discrepancy, of 18.5%. A similar outcome is likewise seen for strokes 2 and 4.

Figure 8. Cycle-averaged velocity magnitude profiles of Line 1 for SPH, SPH D-S and FV methods

The cycle-averaged velocity magnitude profiles across Line 2 (in the middle of the baffled cell, see Figure 3) are shown in Figure 9 for the FV, SPH and SPH D-S approaches. It is seen that the velocity profiles at the strokes 1 and 3 by the FV not only differ in velocity amplitude with a maximum error of 8.7%, but also differ in shape. Conversely, the velocity profiles by the SPH D-S method display repeatable cyclic flow with a maximum error of 0.5% at the same strokes. It demonstrates again that a density-smoothing method is essential to provide regular and predictable flow patterns with SPH, showing good reproducibility and consistency for modelling oscillatory transient problems.

For strokes 2 and 4 representing the end of both the forward and backward strokes respectively, the comparisons of velocity profiles are better for both the FV and SPH D-S methods; however the magnitudes for the former are larger than for the latter.

These differences in the magnitude and shape of velocity profiles among FV strokes, and between FV and SPH D-S, at Line 2, are due to the earlier mentioned phenomenon: Figure 7 showed that while eddy structures generated by SPH D-S remain close to the walls of the baffles, eddies observed in FV solutions are slightly displaced towards the center of the baffled cell. This leads to an increment in velocity magnitude at Line 2, in comparison with those obtained with SPH D-S. This eddy displacement shown in FV results occurs at every cycle; the distance travelled by the eddies remains significantly constant throughout cycles, leading to a quasi-steady-state, hence a stable cycle-averaged velocity field is obtained (Figure 7). However, subtle alterations in the distances travelled by eddies from cycle to cycle manifest into significant differences in the magnitude of velocity and the shape of velocity profiles at Line 2 among FV strokes. This is clearly observed in Figure 11, where velocity magnitude profiles, obtained with FV and extracted at Line 2, are presented at strokes 1 to 4 for different oscillatory cycles. Likewise, Figure 12 shows the analogous for SPH D-S. Undoubtedly, SPH D-S does a better job at producing cycle-repeatable results than FV, especially for velocity profiles at a vertical centered line (at the middle of the baffled cell). Nonetheless, despite this phenomenon and its impact on velocity profiles at Line 2, both methodologies present very similar cycle-repeatable velocity fields across the entire domain, capturing almost identical eddy generation patterns, as stated in section 4.3.1.

Figure 9. Cycle-averaged velocity magnitude profiles of Line 2 for SPH, SPH D-S and FV methods

The cycle-averaged velocity magnitude profiles shown in Figure 10 are measured across a horizontal center line (Line 3 in Figure 3). Again, the standard SPH approach suffers from noise arising from its density distribution, consequently, the velocity magnitudes for all strokes display small fluctuations that have a dampening effect on the overall flow profile, in particular in strokes 2 and 4 where the velocity components are small. These density fluctuations and the associated velocity dampening effect are eliminated by the use of SPH D-S. As earlier mentioned, the numerical solution for strokes 1 and 3, and strokes 2 and 4, should ideally yield a profile that mirrors one another; this is especially expected at Line 3, as it is positioned across the axial direction. Both FV and SPH D-S methodologies successfully

predict mirror-shaped velocity profiles among strokes, and show good agreement and resemblance among one another. Local differences in velocity magnitude are the results of the aforementioned eddy displacement produced by FV.

Figure 10. Cycle-averaged velocity magnitude profiles of Line 3 for SPH, SPH D-S and FV methods

Figure 11. Velocity magnitude profiles of Line 2 obtained with Finite Volume (FV) method for different oscillatory cycles

Figure 12. Velocity magnitude profiles of Line 2 obtained with SPH Density-Smoothing (SPH D-S) method for different oscillatory cycles

4.4. Mixing assessment

Diverse methods and indexes have been used in the past in order to quantitatively assess mixing efficiency in OBR, for example, the velocity ratio (Fitch et al., 2005; Jian & Ni, 2005; Manninen et al., 2013) and stretch rates (Mackley & Neves Saraiva, 1999; Roberts & Mackley, 1995). Simulated data obtained with the SPH can directly be utilized to do the same analysis.

4.4.1. Velocity ratio

The previously used axial to radial velocity ratio (R_V) is defined as:

$$R_V(t) = \frac{\sum_{i=1}^{N_T} \left(|u_{x,i}| \frac{m_i}{\rho_i} \right)}{\sum_{i=1}^{N_T} \left(|u_{y,i}| \frac{m_i}{\rho_i} \right)} \quad (26)$$

where $u_{x,i}$ and $u_{y,i}$ are, respectively, the axial and radial velocity components of a particle i , and N_T is the total number of fluid particles in the domain. By directly utilizing the velocity data generated by the SPH D-S method, the cycle-averaged velocity ratio is obtained and plotted against the period of an oscillatory cycle in Figure 13; the shape of which is the same as that reported by Jian and Ni (Jian & Ni, 2005). Past work (Fitch et al., 2005; Jian & Ni, 2005; Manninen et al., 2013) correlated R_V as a function of oscillatory Reynolds number (Re_o) for fluids of different viscosities in OBRs, where an inversely proportional relationship was established for $Re_o < 1000$. The averaged R_V value of 3.18 in this work is slightly higher than the aforementioned work (Jian & Ni, 2005), due to the lower oscillatory Reynolds number ($Re_o = 471$) under which the current study was performed.

Figure 13. Cycle-averaged velocity ratio for the period of an oscillatory cycle

4.4.2. Stretch rates

Another way of quantifying the rates of mixing in a system, due to its fluid dynamics, is by analysing the rates at which infinitesimal lines of the fluid domain are stretched (Ottino, 1989). This approach requires Lagrangian tracking of these infinitesimal lines that compose the domain under evaluation. Advantageously, by modelling the flow field with SPH, the necessity of superimposing a Lagrangian tracer whose movement is integrated based on an Eulerian velocity field (Mackley & Neves Saraiva, 1999; Roberts & Mackley, 1995) is avoided. It is an easy task to assign an infinitesimal line to each fluid particle that comprises the domain in SPH; these lines are then ascribed an initial orientation (at time = 0 s) relative to the axis in the x -direction, defined by the anticlockwise angle θ . The rate of rotation is thus given by:

$$\frac{d\theta}{dt} = \cos^2 \theta \frac{\partial u_y}{\partial x} + \sin \theta \cos \theta \left(\frac{\partial u_y}{\partial y} - \frac{\partial u_x}{\partial x} \right) - \sin^2 \theta \frac{\partial u_x}{\partial y} \quad (27)$$

and the instantaneous exponential stretch rate of an infinitesimal line is defined as:

$$\frac{d(\ln l)}{dt} = \sin^2 \theta \frac{\partial u_y}{\partial y} + \sin \theta \cos \theta \left(\frac{\partial u_y}{\partial x} + \frac{\partial u_x}{\partial y} \right) + \cos^2 \theta \frac{\partial u_x}{\partial x} \quad (28)$$

The time-averaged exponential stretch rate $S(t)$ of a line can be integrated as:

$$S(t) = \frac{1}{t} \int_0^t \frac{d(\ln l)}{dt} dt \quad (29)$$

The above is then volume-averaged for all the lines comprising the system – $\bar{S}(t)$ – and plotted with time, giving an asymptotic value $\bar{S}_{as}$. Exponential stretch rates have units of s^{-1} ,

and are scaled with the frequency of the oscillation (f) in order to obtain dimensionless values. Roberts and Mackley (Roberts & Mackley, 1995) obtained $\bar{S}(t) - Re$ and $\bar{S}(t) - St$ relations, while the effect of Re_o on $\bar{S}(t)$ was later established by Mackley and Neves Saraiva (Mackley & Neves Saraiva, 1999). Based on their findings, $\bar{S}_{as}$ values between 0.5 and 1 are expected for the operational conditions of the current system under evaluation; the asymptotic values in Table 4 fall within the range. Figure 14 shows the evolution of $\bar{S}(t)$ for different initial values of θ ; the profiles and trends are identical to those of the aforementioned research works.

Table 4. Effect of the initial orientation of the infinitesimal lines on the asymptotic value of the systems exponential stretch rate

$\theta_{initial}$	$\bar{S}_{as}$
0°	0.83
45°	0.98
90°	0.97
135°	0.94

Figure 14. *Effect of the initial orientation of the infinitesimal lines on the time evolution of the time-averaged exponential stretch rate*

4.4.3. Distribution of neighboring fluid particles

Using the SPH approach, an alternative mixing assessment can be proposed by defining the domain as two identical immiscible fluids; these fluids can initially be displayed in serial or in parallel, as shown in Figures 15 and 16 (time = 0 s). As the simulation advances with time, visual and qualitative assessment of axial and radial mixing efficiency are observed in Figures 15 and 16, respectively.

Figure 15. *Qualitative axial mixing assessment for SPH D-S and $\Delta x = 0.0005m$*

Figure 16. *Qualitative radial mixing assessment for SPH D-S and $\Delta x = 0.0005m$*

The evolution of mixing efficiency with time can be obtained by quantifying, for each particle, how many of its neighboring particles are of Fluid A, and how many of its neighbors are of fluid B. Numerically, this is done as follows. Let each particle i carry a dimensionless binary variable J_i , whose value can be 0 (if it is a Fluid A particle) or 1 (Fluid B). By doing so, a time-dependent dimensionless “neighboring mixing index”, $NM(t)$, is defined for each particle as:

$$NM_i(t) = \sum_{j=1}^N 2 \frac{m_j}{\rho_j} |J_i - J_j| \tilde{W}_{ij} \quad (30)$$

where the kernel W_{ij} has been corrected using the Shepard filter (Panizzo, 2004), in order to avoid irregularities with particles close to the boundaries, as those will have a lower number of neighboring fluid particles than particles in the bulk. $\tilde{W}_{ij}$ is defined as:

$$\tilde{W}_{ij} = \frac{W_{ij}}{\sum_{j=1}^N \frac{m_j}{\rho_j} W_{ij}} \quad (31)$$

The neighboring mixing index is averaged over all the particles comprising the fluid domain. At time zero, every Fluid A particle has a certain number of neighboring particles, all of which are of Fluid A; likewise, every Fluid B particle only has Fluid B neighbors. This will be true for all particles, except those at the interface that separates the two types of fluids, hence yielding a $NM(t=0s) \approx 0$. On the contrary, a fully mixed system is achieved when each particle has an equal number of neighbors of Fluid A and Fluid B, resulting in $NM = 1$.

Figure 17. Effect of the initial display of fluids A and B on the time evolution of NM for $\Delta x = 0.001m$ (left) and $\Delta x = 0.00025m$ (right)

Figure 17 plots the NM as a function of time for both fluid placements as shown in Figures 15 and 16, for two initial particle distribution spacing (Δx). It is seen that both the axial and radial neighboring mixing indices increase with time and level off at about 0.6 (Figure 17 left). This is expected, as there is no interaction between eddies at the selected operational conditions. Using a smaller Δx (higher resolution), increments of both axial and radial

neighboring mixing indices with time are slower, as the domain is composed by a larger number of particles, taking longer times for them to intermix. Figure 17 also shows a better mixing performance in the radial than in the axial direction as the asymptotic value is reached faster when fluids A and B are initially displayed in parallel. Using the rate at which NM changes with time (i.e. dNM/dt) during the first oscillatory cycles for both initial fluid placements, a new index for plug flow (PF) can be defined as:

$$PF = \frac{\left(\frac{dNM}{dt}\right)_{axial}}{\left(\frac{dNM}{dt}\right)_{radial}} \quad (32)$$

The lower the value of PF , the better the plug flow behavior the system can achieve. The slopes in Figure 17 together with PF values are listed in Table 5 where a small PF value (0.31) is consistently obtained regardless of Δx , indicating near plug flow performance. Note that computational time constrictions prevented simulations with small Δx values from running long enough for an asymptotic $NM(t)$ value to be achieved. However, for the sake of demonstration, the results for $\Delta x = 0.001m$ are presented here, which accurately predict the ratio of axial to radial NM change with time.

Table 5. Change rates of NM and PF values for different flow field resolutions (Δx)

Δx (m)	$(dNM/dt)_{axial}$	$(dNM/dt)_{radial}$	PF
0.001	0.0084	0.0274	0.31
0.00025	0.0219	0.0718	0.31

It should be noted that a more complex and computationally expensive Eulerian-Lagrangian coupled solver (Discrete Phase Model) is required by the FV methodology to provide similar information of individual particles as the one obtained with the SPH.

5. Conclusions

In this paper, SPH, a relatively new Lagrangian approach, has successfully been implemented and utilized for the first time to model and predict symmetrical flow patterns and to assess mixing efficiency in a 2-dimensional OBR system.

The SPH has effectively captured the expected flow characteristics in an oscillatory baffled reactor and produced clear higher velocity regions at the baffle constriction during strokes 1 and 3, and eddy formation during the change in direction at strokes 2 and 4. The density-smoothing function in SPH is important to offset density fluctuations stemmed from the weakly compressible model. The results from SPH D-S provide a more consistent quasi-steady-state flow and show a higher degree of cycle-repeatability than that from its Eulerian counterpart.

An added advantage of SPH is that it allows quantitative assessments of mixing without the need for additional models like Eulerian based methods, due to its readily available information of individual fluid particles. This work has not only demonstrated its potential to easily implement the existing methods to quantify mixing, such as velocity ratio and stretch rates, but also proposed new indexes for assessing mixing and plug flow efficiency by making full use of SPH's capabilities. The proposed SPH methodology has great potential for modelling flows when two phases are involved, e.g. solids in liquid in crystallization processes, as the flow in SPH is driven by particle-particle interaction, allowing for the

implementation of new physics based on these interactions; successfully modelling single phase flow is an essential first step forward for multiphase cases.

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